

Name _____

**ME301 – Applications of
Thermodynamics**

Circle section: 01 [4th Mech] 02 [5th Mech]
 03 [4th Thom] 04 [5th Thom]
 05 [4th Lui]

Exam 2

Oct 26, 2017

Rules:

- Closed book/notes exam.
- Two help sheets allowed. (8-1/2 x 11" sheet of paper, one side, handwritten; may not contain worked out example problems)
- Maple, Excel, and MATLAB are allowed on your laptop, but nothing may be prepared before the exam.
- Either open property tables (hardcopies) or open Property Calculator Programs on your laptop.

Instructions:

- Show all work for complete credit.
- Work in symbols first, plugging in numbers and performing calculations last.

+1 for CM

Problem 1	_____ / 32
Problem 2	_____ / 37
Problem 3	_____ / 30
Total	_____ / 100

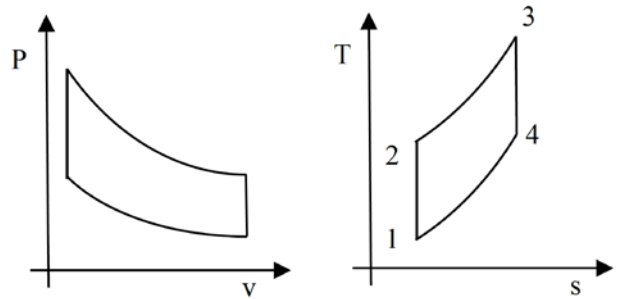
PROBLEM 1 [32 points]

(a) [14 pts] Indicate whether each of the following statements is true or false.

- True | False** One of the air standard assumptions is to model an open system cycle as a closed system.
- True | False** Increasing compression ratio increases the efficiency of an Otto cycle.
- True | False** The specific entropy of a fluid exiting a steady-state, adiabatic turbine increases as turbine efficiency increases.
- True | False** Heat pump and refrigeration cycles form clockwise plots on a T - s diagram.
- True | False** Power cycles form (heat engines) form clockwise plots on a P - v diagram.
- True | False** According to Dalton's model, mixtures of ideal gases can also be assumed to be ideal gases.
- True | False** Partial pressure is calculated by multiplying mass fraction with total pressure.

(b) [7 pts] Given the P - v and T - s diagrams at the right,

- add state numbers to the P - v diagram that correspond with the states of the T - s diagram.
- Show where the new state point 2 would be on each diagram if the compression ratio were increased while keeping the initial volume the same.



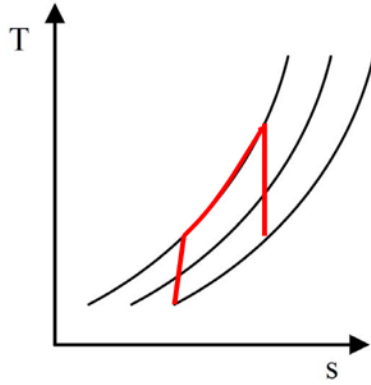
(c) [2 pts] When finding specific entropy changes for one component of an ideal gas mixture, what pressure should you use?

- Mixture pressure
- Partial pressure of the component
- It doesn't matter, either one will always work.
- None of the above – specific entropy of an ideal gas is a function of temperature only

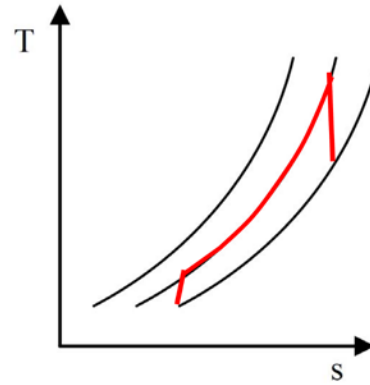
(d) [3 pts] Match the component of a refrigeration cycle with the correct description.

- | | |
|----------------------|---|
| ___ Throttling valve | i. can never be reversible |
| ___ Compressor | ii. relates to refrigeration capacity |
| ___ Condenser | iii. relates to the denominator of the coefficient of performance |

- (e) [6 pts] Compare the T - s diagrams of two gas turbine cycles (two Brayton cycles) as shown. Making use of *cold air standard assumptions*, indicate which cycle better fits each description.



Cycle A



Cycle B

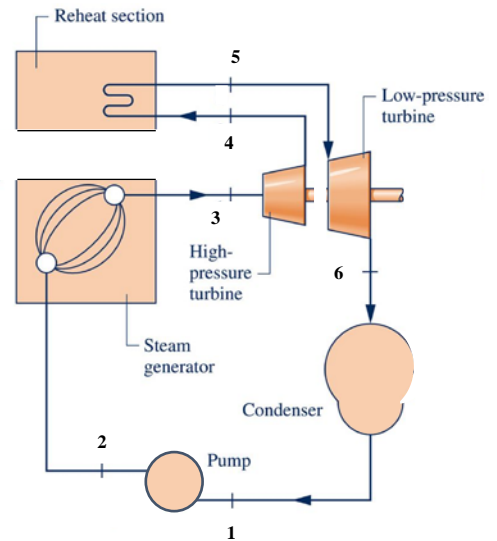
Circle the better choice

- | | | |
|---|---|---|
| i. Has a higher pressure ratio: | A | B |
| ii. Has a higher turbine inlet temperature: | A | B |
| iii. Has a higher cycle efficiency: | A | B |
| iv. Could makes use of a regenerator: | A | B |

PROBLEM 2 [37 points]

A Rankine cycle is modified with reheat as shown in the figure. The flowrate of steam is $\dot{m} = 1.5 \text{ kg/s}$. The isentropic efficiencies of the pump and low-pressure turbine are $\eta_p = 0.7$ and $\eta_{HPT} = 0.8$, respectively. Other operating conditions are given in the figure, and some property information is given in the table below.

State	P [kPa]	T [°C]	x	v [m ³ /kg]	u [kJ/kg]	h [kJ/kg]	s [kJ/kg·K]
1		20.0	0.0	0.001018	83.83	83.84	0.2962
2s	8000	20.1		0.001018	83.83	91.83	
2	8000						
3	8000	700		0.0548	3443	3881	7.280
4s	1000			0.2796	2865	3145	
4	1000						7.506
5	1000	700		0.4478	3476	3924	8.274
6s			0.95	55.07	2294	2422	
6				61.55	2429	2573	8.783



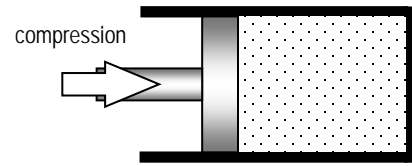
$$\dot{Q}_{generator} = 5679 \text{ kW}$$

$$\dot{W}_{LPT} = 2027 \text{ kW}$$

- Find the inlet pressure to the pump in kPa.
- Calculate the power into the pump in kW.
- Calculate the power out of the high-pressure turbine in kW.
- Calculate the rate of heat transfer into the reheat section in kW.
- Calculate the cycle efficiency.
- Calculate the isentropic efficiency of the low-pressure turbine.

Problem 3 [30 points]

Two kilograms of a two-component mixture of 20% of methane (CH_4) and 80% of oxygen (O_2) by mass is compressed **isothermally** in a closed piston cylinder from an initial pressure and temperature of $P_1 = 100 \text{ kPa}$ and $T_1 = 25^\circ\text{C}$ to a final pressure of $P_2 = 500 \text{ kPa}$.



Treat the mixture as an ideal gas with variable specific heats. For reference, the molar masses of methane and oxygen are $M_{\text{CH}_4} = 16.04 \text{ kg/kmol}$ and $M_{\text{O}_2} = 32.00 \text{ kg/kmol}$, respectively, and $\bar{R} = 8.314 \text{ kJ/kmol}\cdot\text{K}$.

- (a) Determine the molar composition (i.e., the mole fractions) of the mixture.
- (b) Sketch the compression process on both P - v and T - s diagrams.
- (c) Find the following properties and changes in properties:
 - i. R_{mix} in $\text{kJ/kg}\cdot\text{K}$
 - ii. v_{1,O_2} in m^3/kg
 - iii. $(\bar{h}_2 - \bar{h}_1)_{\text{CH}_4}$ in kJ/kmol
 - iv. $(\bar{s}_2 - \bar{s}_1)_{\text{mix}}$ in $\text{kJ/kmol}\cdot\text{K}$

Length

$$1 \text{ ft} = 12 \text{ in} = 0.3048 \text{ m} = 1/3 \text{ yd}$$
$$1 \text{ m} = 100 \text{ cm} = 1000 \text{ mm} = 39.37 \text{ in} = 3.2808 \text{ ft}$$
$$1 \text{ mile} = 5280 \text{ ft} = 1609.3 \text{ m}$$

Mass

$$1 \text{ kg} = 1000 \text{ g} = 2.2046 \text{ lbm}$$
$$1 \text{ lbm} = 16 \text{ oz} = 0.45359 \text{ kg}$$
$$1 \text{ slug} = 32.174 \text{ lbm}$$

Temperature Values

$$(T/\text{K}) = (T/^\circ\text{R}) / 1.8$$
$$(T/\text{K}) = (T/^\circ\text{C}) + 273.15$$
$$(T/^\circ\text{C}) = [(T/^\circ\text{F}) - 32] / 1.8$$
$$(T/^\circ\text{R}) = 1.8(T/\text{K})$$
$$(T/^\circ\text{R}) = (T/^\circ\text{F}) + 459.67$$
$$(T/^\circ\text{F}) = 1.8(T/^\circ\text{C}) + 32$$

Temperature Differences

$$(\Delta T/^\circ\text{R}) = 1.8(\Delta T/\text{K})$$
$$(\Delta T/^\circ\text{R}) = (\Delta T/^\circ\text{F})$$
$$(\Delta T/\text{K}) = (\Delta T/^\circ\text{C})$$

Volume

$$1 \text{ m}^3 = 1000 \text{ L} = 10^6 \text{ cm}^3 = 10^6 \text{ mL} = 35.315 \text{ ft}^3$$
$$= 264.17 \text{ gal}$$
$$1 \text{ ft}^3 = 1728 \text{ in}^3 = 7.4805 \text{ gal} = 0.028317 \text{ m}^3$$
$$1 \text{ gal} = 0.13368 \text{ ft}^3 = 0.0037854 \text{ m}^3$$

Volumetric Flow Rate

$$1 \text{ m}^3/\text{s} = 35.315 \text{ ft}^3/\text{s} = 264.17 \text{ gal}/\text{s}$$
$$1 \text{ ft}^3/\text{s} = 1.6990 \text{ m}^3/\text{min} = 7.4805 \text{ gal}/\text{s}$$
$$= 448.83 \text{ gal}/\text{min}$$

Force

$$1 \text{ N} = 1 \text{ kg m}/\text{s}^2 = 0.22481 \text{ lbf}$$
$$1 \text{ lbf} = 1 \text{ slug ft}/\text{s}^2 = 32.174 \text{ lbm ft}/\text{s}^2$$
$$= 4.4482 \text{ N}$$

Pressure

$$1 \text{ atm} = 101.325 \text{ kPa} = 1.01325 \text{ bar}$$
$$= 14.696 \text{ lbf}/\text{in}^2$$
$$1 \text{ bar} = 100 \text{ kPa} = 10^5 \text{ Pa}$$
$$1 \text{ Pa} = 1 \text{ N}/\text{m}^2 = 10^{-3} \text{ kPa}$$
$$1 \text{ lbf}/\text{in}^2 = 6.8947 \text{ kPa} = 6894.7 \text{ N}/\text{m}^2 ;$$
$$[\text{lbf}/\text{in}^2 = \text{“psi”}]$$

Energy

$$1 \text{ J} = 1 \text{ N m}$$
$$1 \text{ kJ} = 1000 \text{ J} = 737.56 \text{ ft} \cdot \text{lbf} = 0.94782 \text{ Btu}$$
$$1 \text{ Btu} = 1.0551 \text{ kJ} = 778.17 \text{ ft} \cdot \text{lbf}$$
$$1 \text{ ft} \cdot \text{lbf} = 1.3558 \text{ J}$$
$$1 \text{ J} = \text{Pa m}^3$$

Energy Transfer Rate

$$1 \text{ kW} = 1 \text{ kJ}/\text{s} = 737.56 \text{ ft} \cdot \text{lbf}/\text{s} = 1.3410$$
$$\text{hp} = 0.94782 \text{ Btu}/\text{s}$$
$$1 \text{ Btu}/\text{s} = 1.0551 \text{ kW} = 1.4149 \text{ hp}$$
$$= 778.17 \text{ ft} \cdot \text{lbf}/\text{s}$$
$$1 \text{ hp} = 550 \text{ ft} \cdot \text{lbf}/\text{s} = 0.74571 \text{ kW}$$
$$= 0.70679 \text{ Btu}/\text{s}$$

Specific Energy

$$1 \text{ kJ}/\text{kg} = 1000 \text{ m}^2/\text{s}^2$$
$$1 \text{ Btu}/\text{lbm} = 25037 \text{ ft}^2/\text{s}^2$$
$$1 \text{ ft} \cdot \text{lbf} / \text{lbm} = 32.174 \text{ ft}^2/\text{s}^2$$